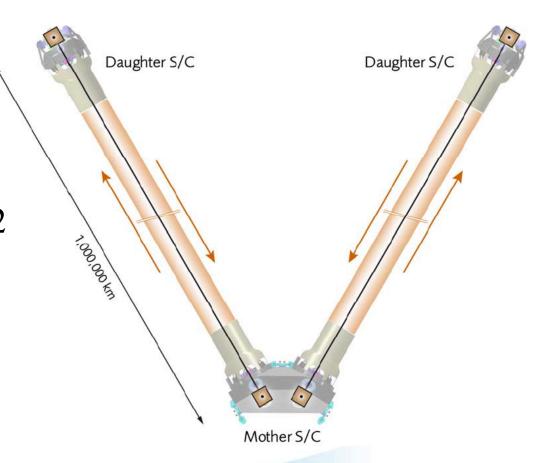




#### NGO



- A space-borne laser interferometer GW astrophysical observatory
- Studied by ESA as a candidate for the selection of a Large mission in 2012





#### An astrophysical observatory



#### • Survey compact stellar-mass binaries and study the structure of the Galaxy

- Elucidate the formation and evolution of Galactic stellar-mass compact binaries and thus constrain the outcome of the common envelope phase and the progenitors of (type Ia) supernovae.
- Determine the spatial distribution of stellar mass binaries in the Milky Way
- Improve our understanding of white dwarfs, their masses, and their interactions in binaries, and enable combined gravitational and electromagnetic observations

#### Trace the formation, growth and merger history of massive black holes.

- − Trace the formation, growth and merger history of massive black holes with masses  $10^5 \, M_{\odot} 10^7 \, M_{\odot}$  during the epoch of growth of quasi-stellar objects and widespread star formation (0 < z < 5) through their coalescence in galactic halos.
- − Capture the signal of coalescing massive black hole binaries with masses  $2 \times 10^4 \, M_{\odot} 10^5 \, M_{\odot}$  in the range of 5 < z < 10 when the universe is less than 1 Gyr old.

#### • Explore stellar populations and dynamics in galactic nuclei

- Characterize the immediate environment of massive black holes in z < 0.7 galactic nuclei from extreme mass ratio capture signals.
- Discovery of intermediate-mass black holes from their captures by massive black holes.

#### • Confront General Relativity with observations

- Detect gravitational waves directly and measure their properties precisely
- Test whether the central massive objects in galactic nuclei are consistent with the Kerr black holes of General Relativity.
- Perform precision tests of dynamical strong-field gravity.

#### Probe new physics and cosmology with gravitational waves

- Measure the spectrum of cosmological backgrounds, or set upper limits on them in the  $10^{-4}$  Hz  $10^{-1}$  Hz band.
- Search for gravitational wave bursts from cosmic string cusps and kinks.
- Cosmography with standard sirens Maritime Institute 20/12/2011



# The approach





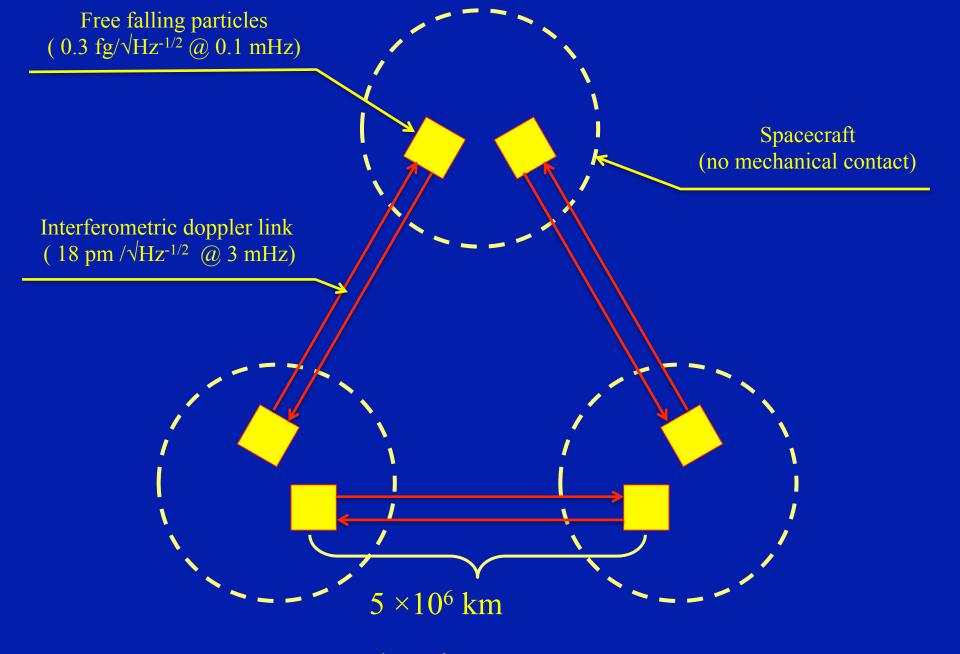
#### Next steps for LISA

The space-borne gravitational wave detector LISA has been studied in great detail over the last decade as a collaborative mission between ESA and NASA.

As announced by ESA in March 2011, due to a modified international cooperation scenario, it is now necessary to study a European-only mission that offers a significant reduction of the cost while maintaining its core science objectives.

Whilst maximizing the use of results of the LISA studies performed so far and the heritage from LISA Pathfinder a number of significant changes to the payload, the spacecraft and the mission architecture have to be made to enable the mission to fit the new budget profile.

• the starting point are LISA with her 6-year long formulation study, and LISA Pathfinder.



# LISA basic concepts

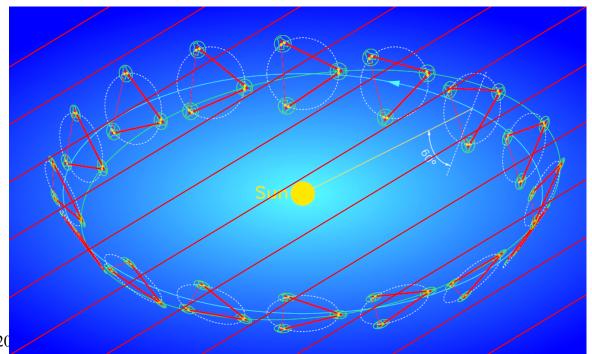
Maritime Institute 20/12/2011 S. Vitale





# LISA basic concept. 1 Orbits

- Formation rotates within waves (pointing capability)
- Constant solar panels illumination
- All laser links within common spacecraft (SC)formation plane





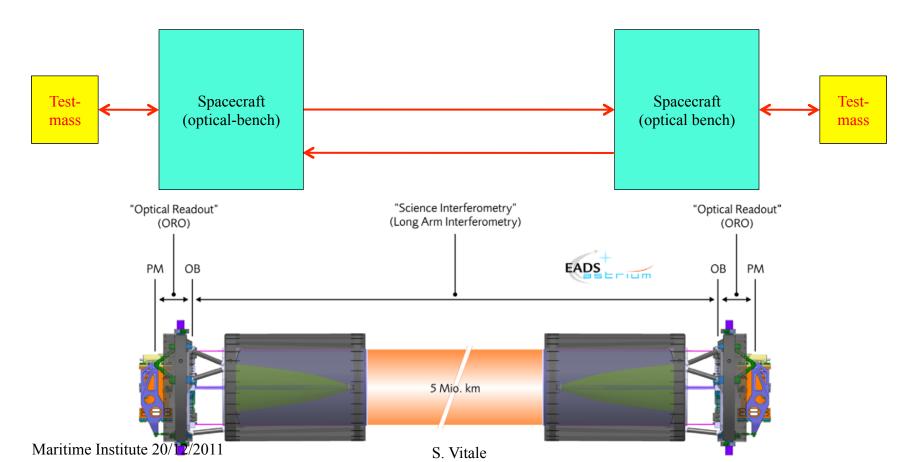
# LISA basic concept 2. The Doppler link of the link of



• A Doppler test-mass (TM) to TM relative speedometer  $\Delta v/v = \Delta v/c$ 



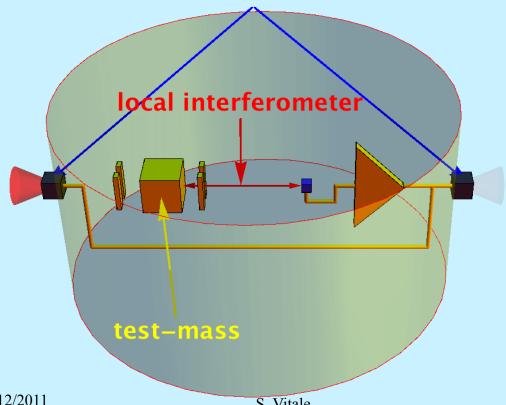
• Real Link is split in three: two TM-SC two-way links, two one-way SC-SC links



# LISA basic concept. 3 Drag-free

- Position of spacecraft relative to test-mass is measured by local interferometer
- Along each link direction, spacecraft is kept centered on test-mass by acting on micro-Newton thrusters.

  Micro-Newton thrusters



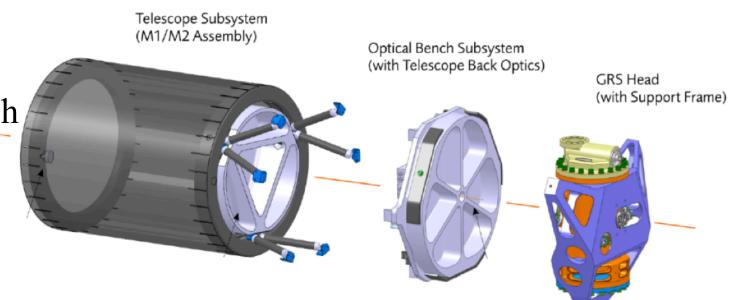


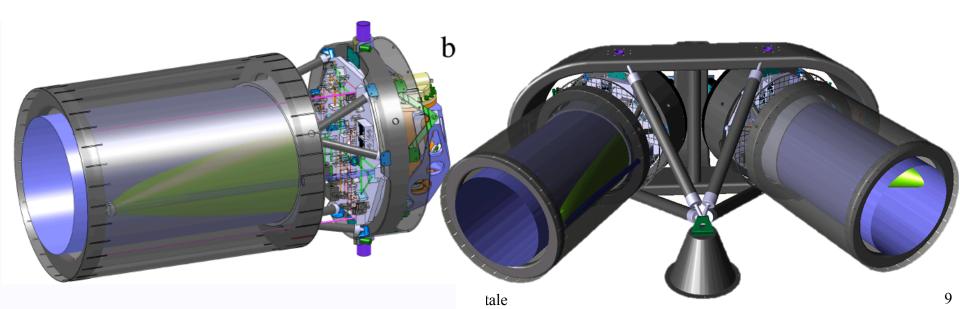
#### INFN Istituto Nazionale di Fisica Nucleare



#### LISA Instrument main elements

- Telescope
- Optical Bench
- GRS
- 2 per SC



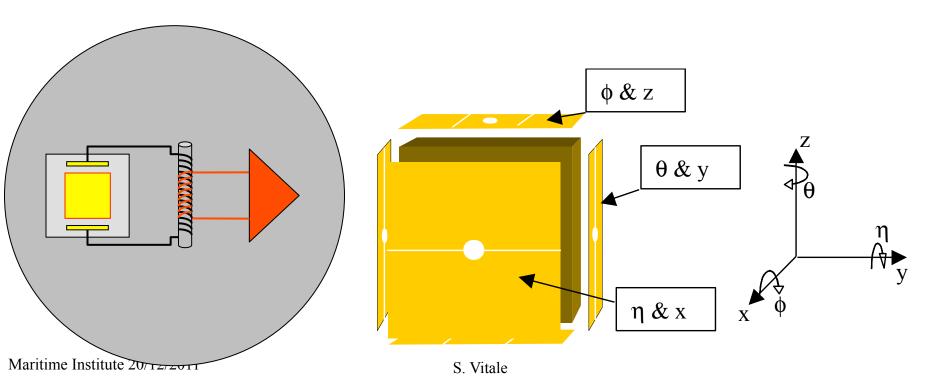




# Gravity Reference Sensor (GRS)

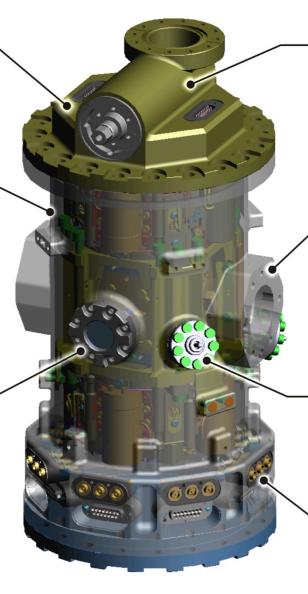


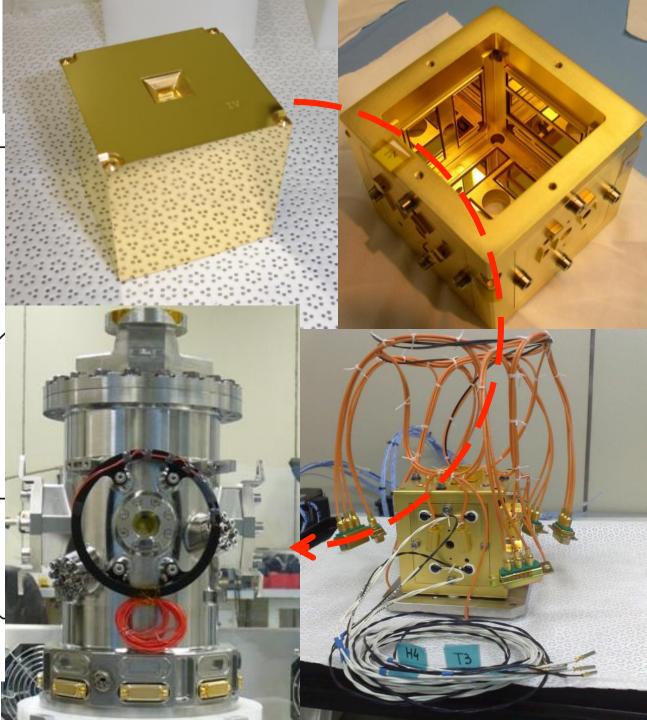
- Provides electrostatic sensing along all degrees of freedom (dof) for initial acquisition and for normal operation of non interferometric dof.
- Provides TM actuation for non-drag-free dof.





# GRS





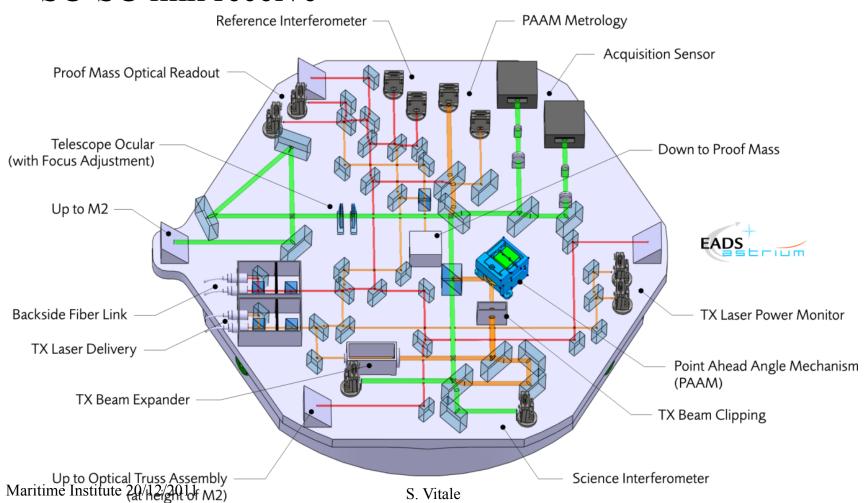
Maritime Institute 20/12/2011



# Optical bench



- Local TM-SC metrology
- SC-SC link send
- SC-SC link receive

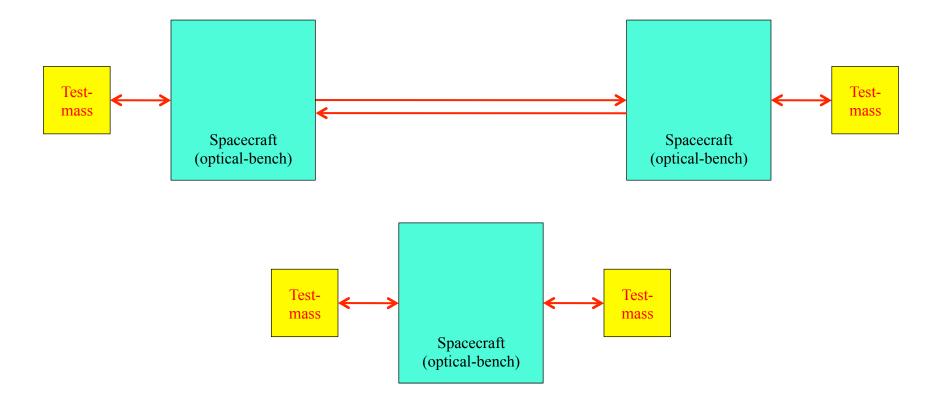






# The Concept of LISA Pathfinder



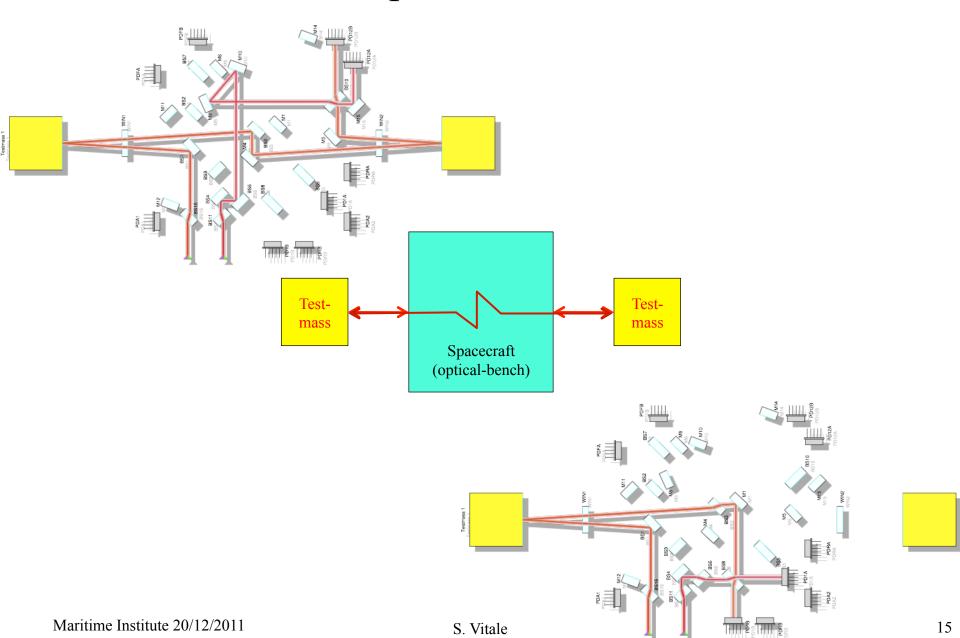


- Drop the 5 Million kilometer spacecraft → spacecraft measurement
- Maintain both spacecraft → test-mass local measurements





# The Concept of LISA Pathfinder



#### Q2: What will LPF test exactly?

- The end-to-end performance of a reduced version of one LISA. arm. Specifically:
- The suppression of parasitic relative acceleration of two test-masses at 3×10<sup>-14</sup> ms<sup>-2</sup>√Hz. (a factor 7 in amplitude from LISA) above 10<sup>-3</sup>Hz (a factor 10 in frequency from LISA)
- The interferometric ranging of two free-falling test-masses at 10pm/√Hz (same as LISA).
- The entire spacecraft to test-mass local measurement, including drag-free and controls, with hardware designed for LISA.
- The detailed physical model of the disturbances, to allow for final verification of the parts of LISA design under test.

Note: LISA Pathfinder results are not needed as an input to the design of LISA. Pathfinder is a verification step within the LISA development.









#### LISA Pathfinder Design and **Development Approach**

- Design is for LISA.
- Relaxation of performance requirements is only allowed to make testing conditions achievable. It cannot affect design.
- Implications:
- When one subsystem item reaches a given level of readiness for LISA Pathfinder (CDR, Pre-launch Review etc.,) it reaches the same level of readiness for LISA too.
- All the development risks up to that stage are consequently retired also for LISA
- Risk of unforeseen system level problems (estimated to be low) is finally retired by the flight.







#### LISA Design Items Verified on Pathfinder (in black)

#### Free flying test mass subject to very low parasitic forces:

- Drag free control of spacecraft (non-contacting spacecraft)
- Low noise microthruster to implement drag-free
- Large gaps, heavy masses with caging mechanism
- High stability electrical actuation on cross degrees of freedom
- Non contacting discharging of test-masses
- Thermomechanical stability of S/C
- Gravitational field cancellation

# Precision interferometric, *local* ranging of test-mass and spacecraft:

- pm resolution ranging, sub-mrad alignments
- High stability monolithic optical assemblies

#### Precision spacecraft to spacecraft ranging:

- High stability telescopes
- High accuracy phase-meter
- High accuracy frequency stabilization
- Constellation acquisition
- Precision attitude control of S/C

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# LPF expected performance



- Interferometry tested endto end in thermal vacuum chamber with entire SC on, and solar radiation simulator
- Performance of interferometers better then requirements: better than 5 pm/√Hz and 0.3 nrad/√Hz at at 10 mHz

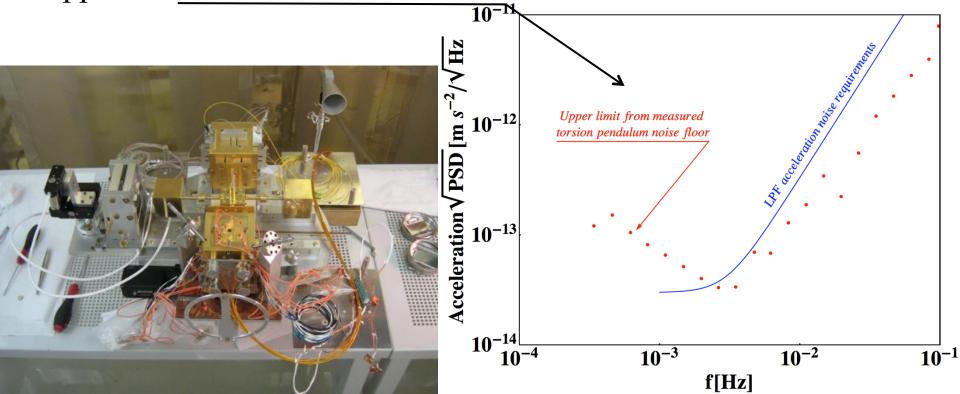






# Torsion pendulum test of GRS performance

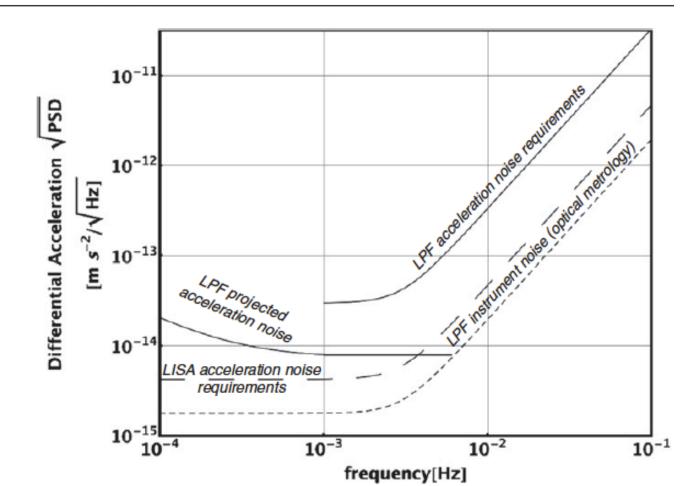
- GRS flight hardware replica tested. Ok for LISA/NGO for:
  - Thermal effects
  - Charge patches noise
  - Actuation noise
  - Molecular impacts
- Upper limit to unforeseen disturbances noise





# Expected LPF performance close to LISA/NGO requirements

Class. Quantum Grav. 28 (2011) 094002

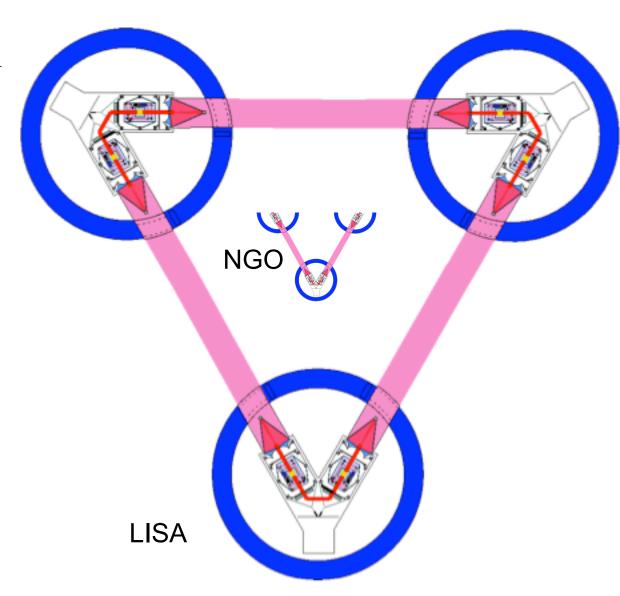




#### NGO



- Key features:
- Suppression of one arm
- Squeezing of size to 1 Mio Km
- Shortening of duration











# Technical simplifications



23

- Reduction of the Arm Length from 5 to 1 Mio. km
  - Enables significant simplification of the Payload
    - Telescope Diameter reduced from 40 cm to 20 cm
    - Laser Power reduced from 2 W to 1.4 W
    - Point-Ahead Angle Mechanism not required
- Realization of only 2 interferometer arms
  - Saves hardware for the 3rd arm
- Duration of Science Operations reduced from 5+5 to 4+2 years
  - Enables a stable "Slow Drift Away" orbit with minimal  $\Delta v$
  - Reduces consumable like propellant for micro-thruster allowing for cold gas

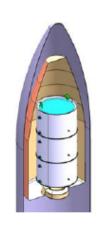


#### Cost reduction measures



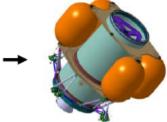


- Dual Soyuz launch scenario
  - Selected on the basis of current ESA cost estimates
  - Instead of single Ariane
- Propulsion modules inherited from LPF
  - Maximum heritage
  - Minimal design changes avoid requalification
- Removal of Ancillary Metrology Functions not required anymore to meet performance requirements
  - no PAAM Metrology
  - no Optical Truss
- Minimal payload height













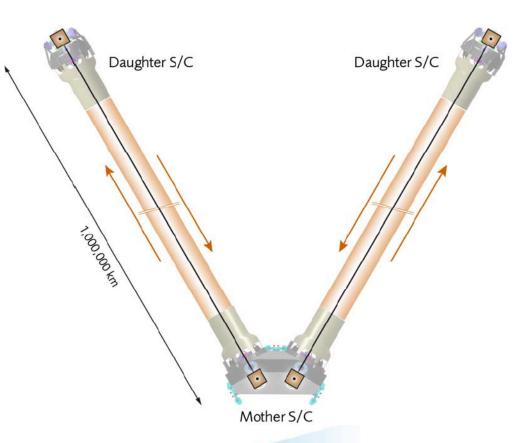


#### Mother/Daughter configuration



• Hardware for the 3rd arm is saved in Daughters. This reduces the Daughter's

- Dry mass
- Propellant mass
- Wet launch mass
- Payload AIVT effort





#### Orbit



- Many considered, Halo, Lagrange points, Earth centered....
- Baseline: slow drift away orbit starting at 10°, reaching 22° after 6 years

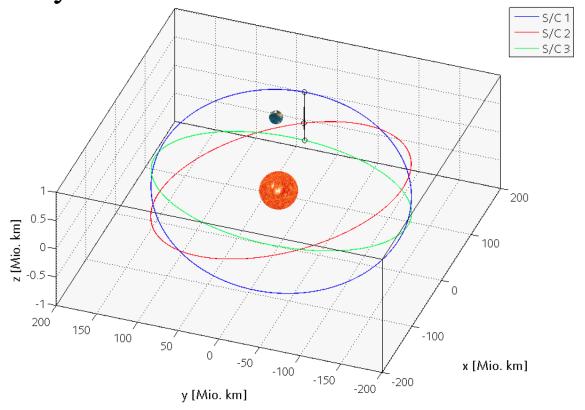


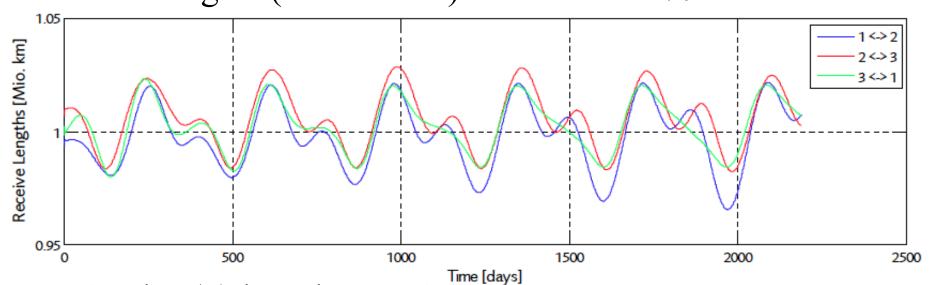
Figure 20: Earth-trailing heliocentric orbits of the 3 individual spacecraft (note the exaggerated scale on the z



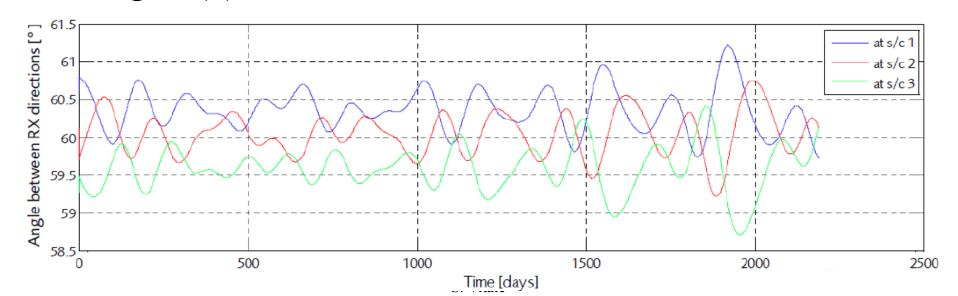
### Arm-length and angles



• Arm-lengths (1-2 and 2-3) stable at 2-3 %



• Angles (2) less then  $0.9^{\circ}$ 

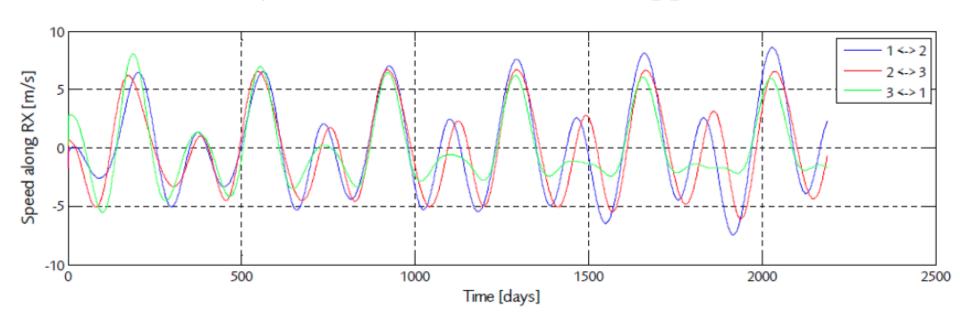




# SC-SC velocity and distance to earth the street and the street and



• Max velocity < 10 Km/s (<10 MHz Doppler shift



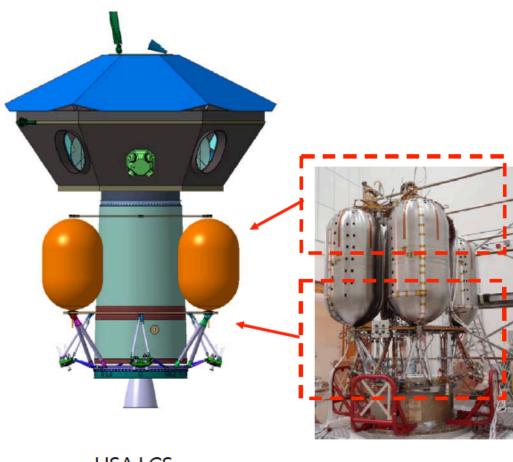
Distance to Earth 30→65 Mio Km in 6 Years





# Launch configuration

- LISA Composite Spacecraft (LCS) consists of Sciencecraft mounted on Propulsion Module
- Propulsion Module derived from LPF by shortening tanks and cone only
  - LPF tanks too large to launch at minimum 70% fill, smaller existing Eurostar family tanks used
  - Identical for Mother and Daughter
- Facetted array permits upper layer inside Soyuz fairing



LISA LCS

LPF PM



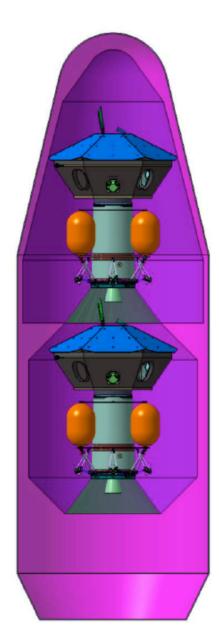
# Launch configuration





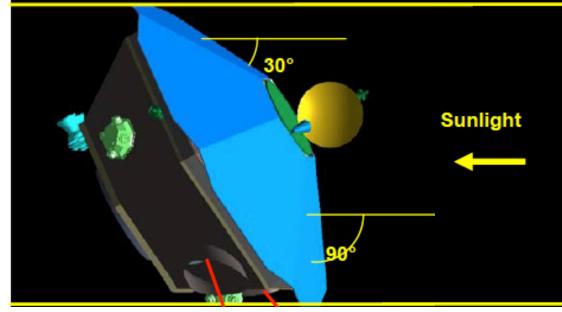
- Daughter and Mother launch look identical
- Mother has useful spare capacity in excess of 1000kg (shared launch)
- Ariane launch also possible, cost less solid
- Budgets. Worst case: RIT power and cold gas mass
- Wet mass
  - Mother 1650 kg
  - Daughter 1583 Kg (each)
- Dry mass (20% margin)
  - Mother 1003 kg
  - Daughter 849 kg
- Payload mass
  - Mother 282 kg
  - Daughter 155 kg
- Power (20% margin)
  - Mother 863 W
  - Daughter 755 W







# Spacecraft



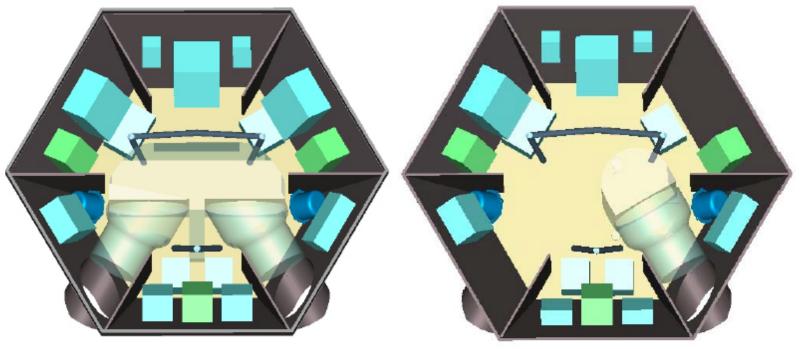


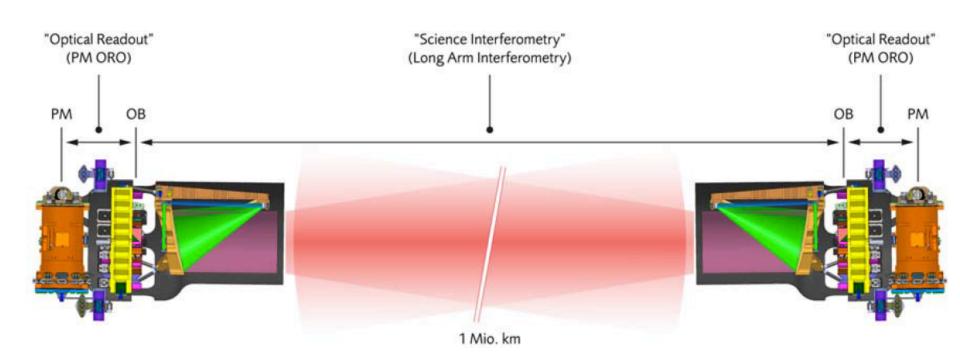
Figure 7: Mother (left) and Daughter (right) payload bays



# Metrology



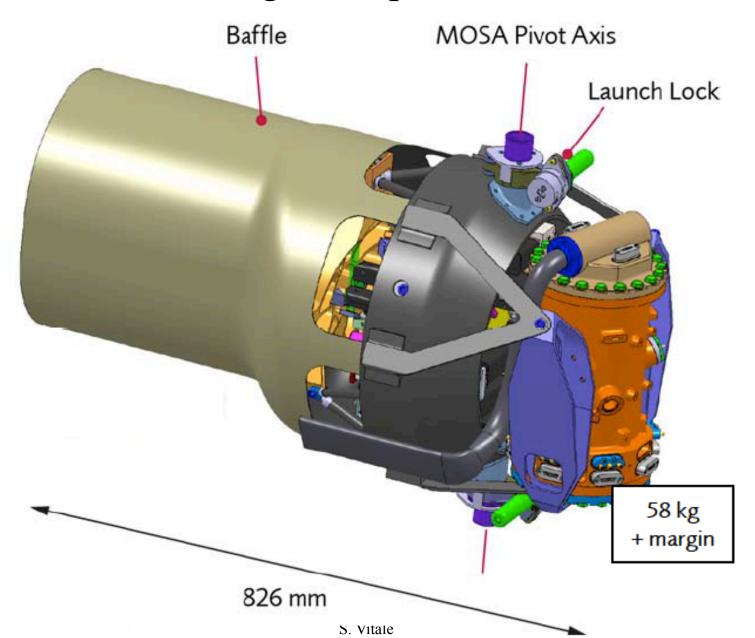
- Link configuration similar to LISA
- GRS identical to LPF
- Metrology simplified by decreased telescope diameter, removal of point ahead mechanism, optical truss







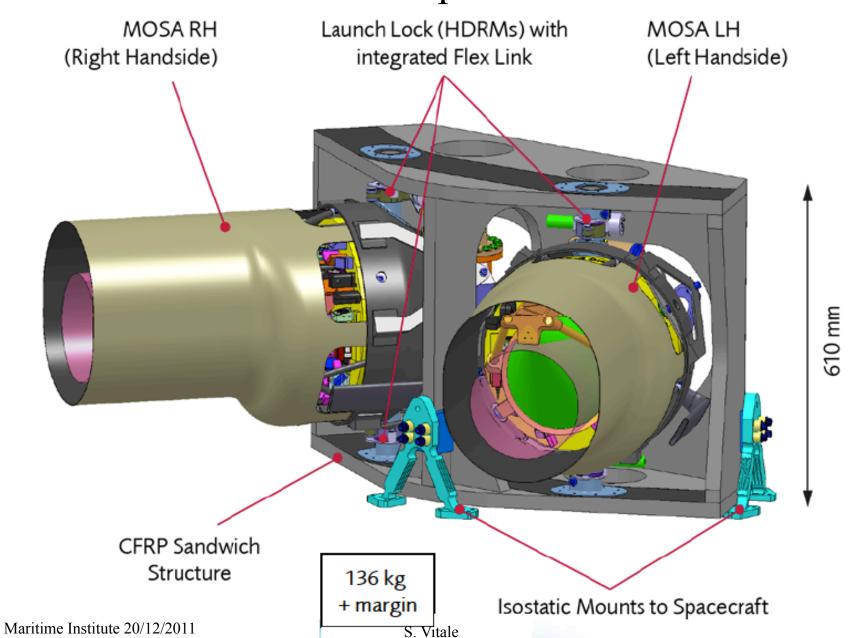
# Daughter Spacecraft







# Mother Spacecraft

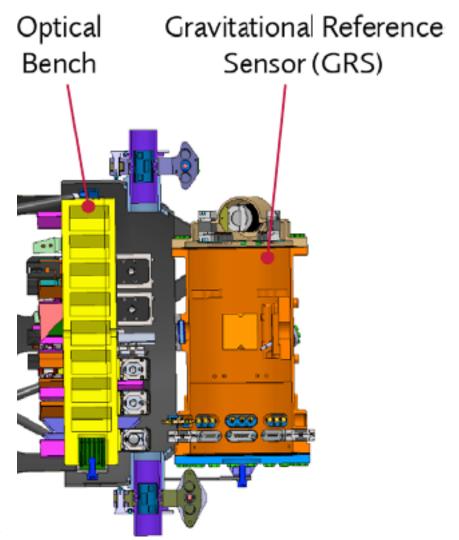




#### Instrument



• The basic complement



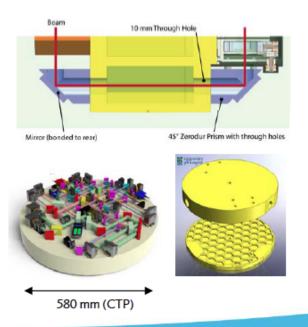
# Astrium, It shall not be communicated to third parties without prior written agreement. Its content shall not be disclose

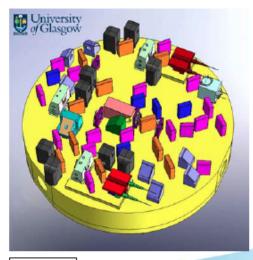
#### Payload Architecture

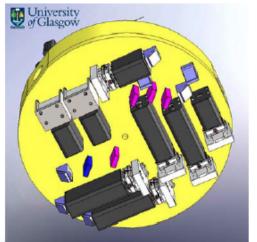
#### Optical Bench (University of Glasgow)



- To take full advantage of reduced telescope dimensions, it is essential to also minimize the mechanical envelope of the OB
  - OB diameter drives payload height (as in LMF), and thus s/c height
  - Enabled by removal of PAAM, PAAM Metrology & Optical Truss on double-sided OB
    - Two sides joint by adhesive bonding (e.g. ultra-stable Astrium precision bonding technology)
    - "Periscopes" for beam transfer









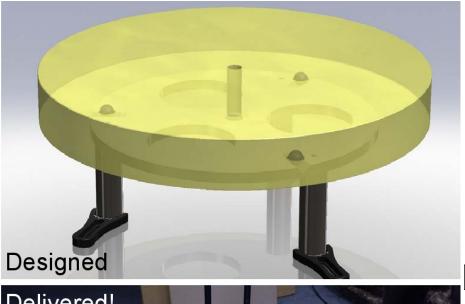




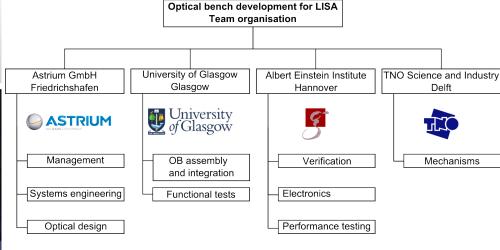
# LISA OB Development (ESA) Control Lista Observation (ESA)











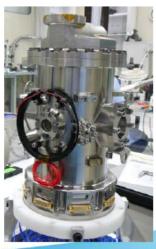
## Payload Architecture

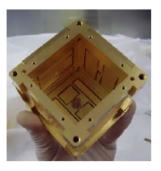
### Gravitational Reference Sensor (GRS)

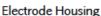


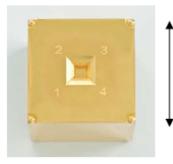
- The Gravitational Reference Sensor provides the inertial reference for the LISA measurement
  - For each MOSA, the inertial reference is established by a "Proof Mass", which is drag-free controlled along the respective measurement direction ("sensitive axis")
  - The GRS sensor heads are actuators and sensors of the DFACS software which is responsible to achieve the drag free condition and to control the telescope pointing
- The architecture of the GRS is directly inherited from the LISA Pathfinder "Inertial Sensor"
  - Minor modifications (apart from lessons learned) have always been anticipated for extrapolation of the LPF results to the specific LISA conditions (e.g. self gravity environment)











AuPt Proof Mass

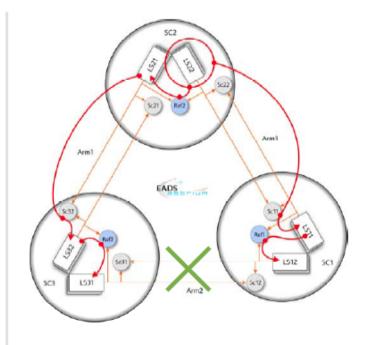


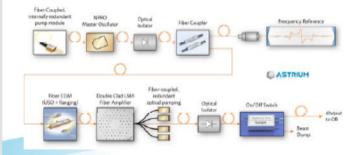
## Payload Architecture

### Laser Subsystem

- Each Optical Bench is fed by a High Power Laser Subsystem (1.1W + 30% margin)
  - Provides laser light for TX Beam
  - Internal cold redundancy between a nominal and redundant Laser Head
  - Frequency Reference for in-orbit frequency stabilization
  - LMF architecture remains applicable
- Each Optical Bench requires additionally a "Local Oscillator" for the local interferometry
  - Obtained from the 2nd OB in the Mother via "Backlink Fiber"
  - Generated by a dedicated Low Power Laser
     Subsystem in the 2 Daughters (~ 50 mW TBC)
    - Copy of Master Oscillator in High Power Laser Subsystems
    - Alternatively can be generated by frequency modulation of High Power laser (cf. LTP)







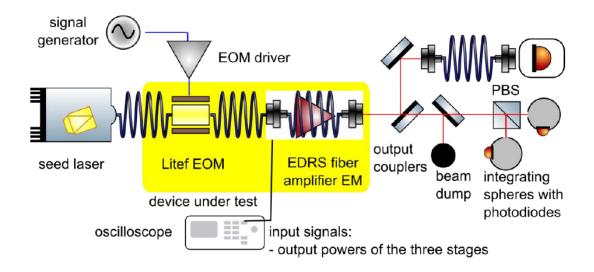


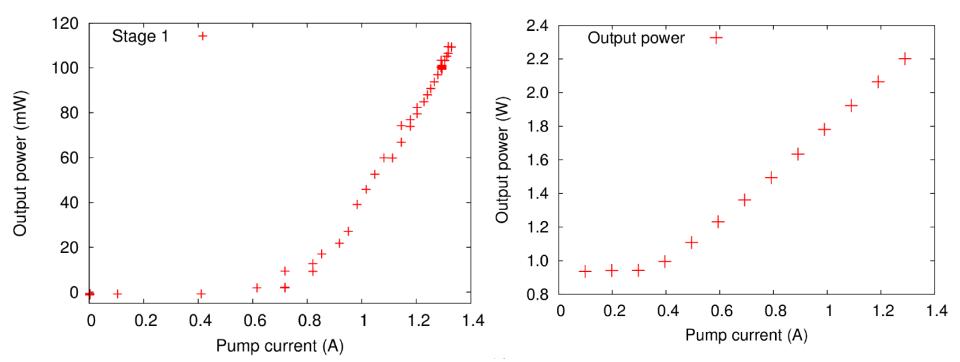


## Laser development in ESA





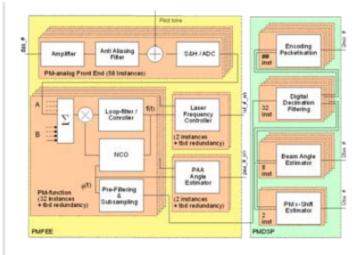




# Payload Architecture Phase Measurement Subsystem

LISA

- The Phase Measurement Subsystem (PMS) is the source of all data used for nominal LISA ground processing with the exception of PM lateral shift (provided from GRS)
  - Measures the **phase** of the beat note of all interferometers (thus providing all lengths and angle measurements determined by interferometry)
  - Provides a consistent datation to these measurements
  - Generates the modulation tones for the USO frequency exchange modulation and measure the phase difference of the sub-carrier beat notes resulting thereof
  - Transmits and receives timing semaphores as needed to measure arm delay and establish a consistent constellation time (evaluation in ground processing)
- Additional functions related to the phase measurements are also allocated to the PMS
  - Provide PM and Pointing metrology to the DFACS
  - Perform Laser Frequency Control
- Architecture and functions identical to LMF
  - Reduced number of channels



	LMF	Mother	Daughter
No. of input lines	70	48	24
No. of active channels	46	32	16
Mass (w/o margin)	16 kg	12 kg	8 kg
Power (w/o margin)	49 W	38 W	23 W





## Under development under ESA contract









LISA Metrology System

Metrology system demonstration model design description and test results Full version ESA ITT Reference Number AO/1-6238/10/NL/HB

TN1-3









LISA Metrology System

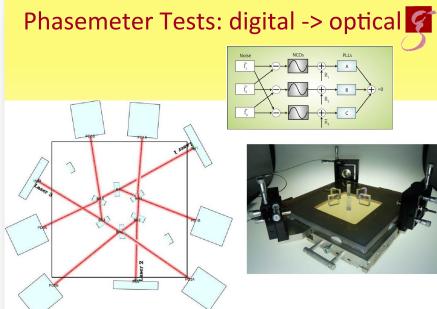
Metrology system consolidated architecture Full version

ESA ITT Reference Number AO/1-6238/10/NL/HB

TN1-2



## Ranging and Data Transfer: Done Free-running Weak- light laser phase-locking Linear 2 GHz + fs PRN 1 breadboard



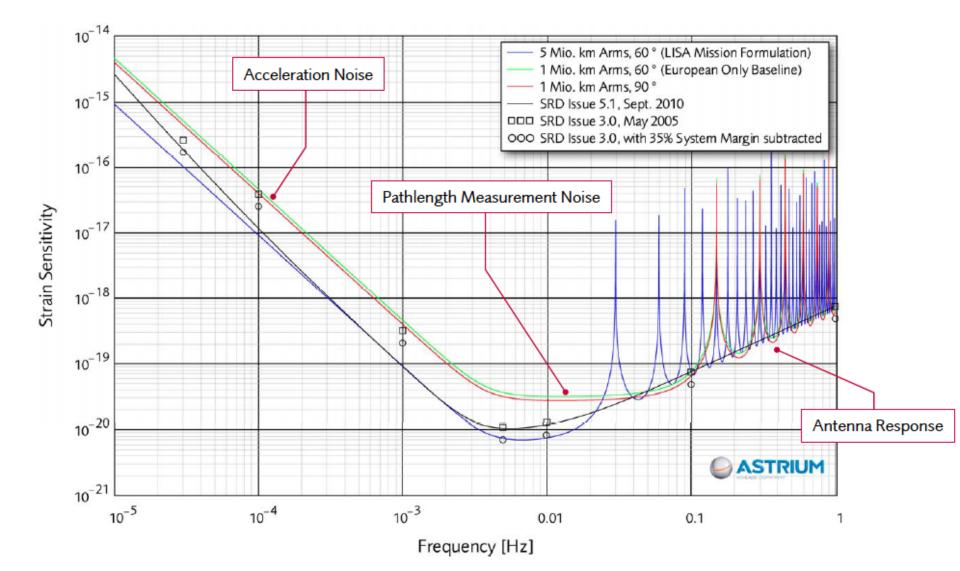


## **Science Performance**

## Strain Sensitivity @ 1 Mio. km Arm Length



43







## Science

#### **Science Objectives**

Through the detection and observation of gravitational waves

- Survey compact stellar-mass binaries and study the structure of the Galaxy
- Trace the formation, growth, and merger history of massive black holes
- Explore stellar populations and dynamics in galactic nuclei
- Confront General Relativity with observations
- Probe new physics and cosmology

#### **Event Rates and Event Numbers**

Frequency band  $1 \times 10^{-4}$  Hz to 1 Hz,  $(3 \times 10^{-5}$  Hz to 1 Hz as a goal)

Massive black hole mergers  $10 \text{ yr}^{-1}$  to  $100 \text{ yr}^{-1}$ 

Extreme mass ratio inspirals  $1 \text{ yr}^{-1}$  to  $10 \text{ yr}^{-1}$ 

Galactic Binaries  $\sim 3000$  resolvable out of a total of  $\sim 30 \times 10^6$  in the NGO band

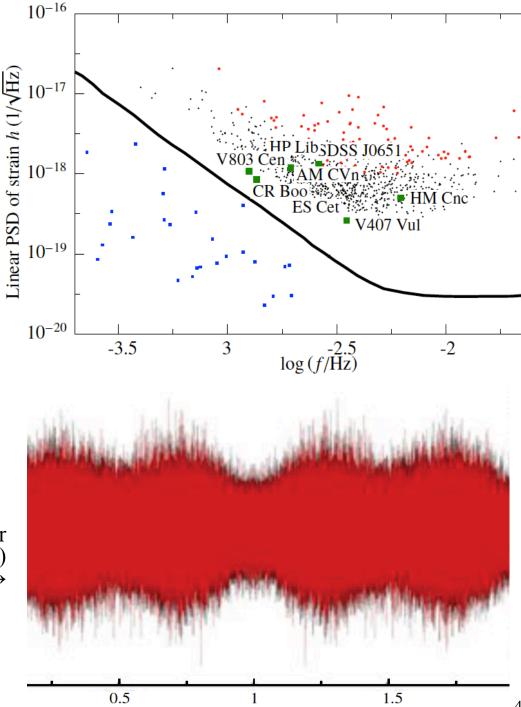
Maritime Institute 20/12/2011



Compact binaries

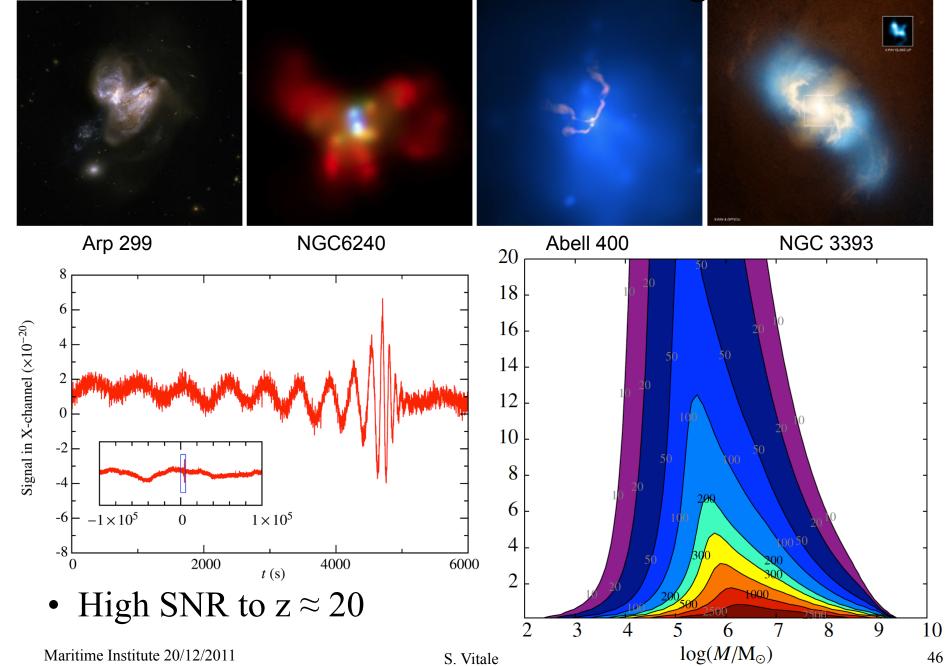
- > 8 Verification binaries (pending upcoming surveys)
- >3000 other individually detectable sources (progenitors of supernovae, test of enhanced formation in globular clusters),
   <3° location, 10° inclination, distances down to 1 %</li>
- Detailed study of high SNR ones give tidal and mass-transfer behavior.
- >10<sup>7</sup> undistinguishable binaries galactic foreground gives population census and properties

signal in time domain before (black) and after (red) subtracting resolvable sources)





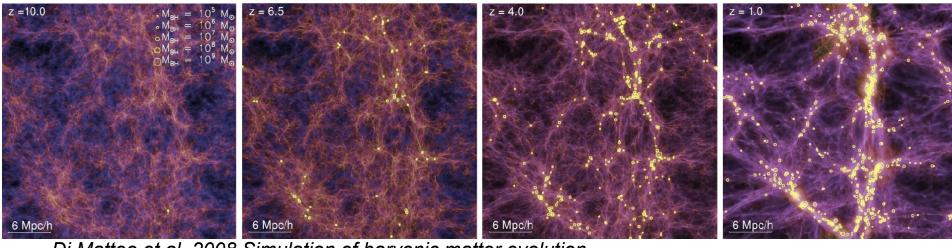






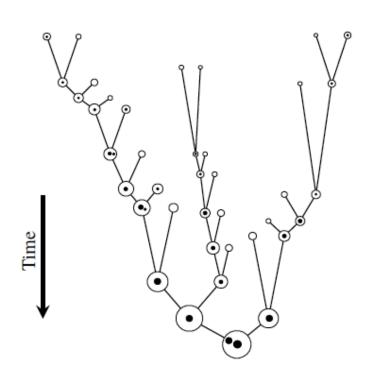
## Cosmic stratigraphy





Di Matteo et al. 2008 Simulation of baryonic matter evolution

- During their cosmic evolution black holes transit into the mass interval to which NGO is sensitive.
- NGO will map galaxies formation and clustering, using black holes as clean tracers.

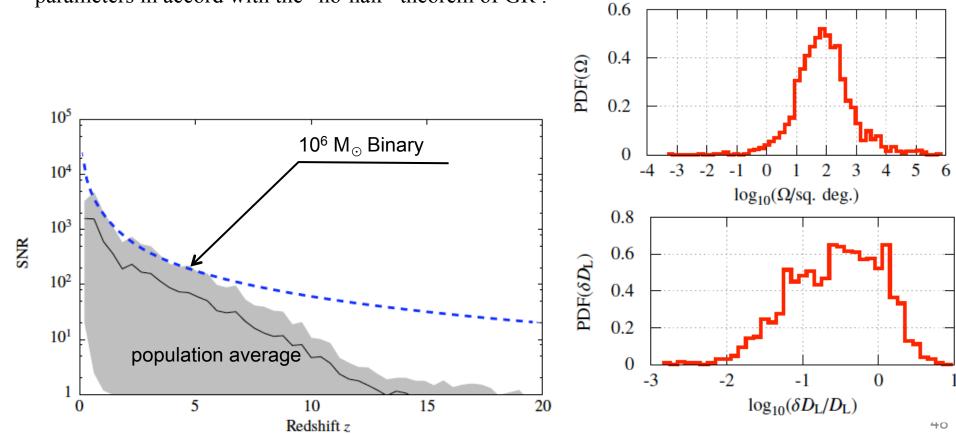




## BH physics at high SNR



- NGO will be able to detect the gravitational waves emitted by black hole binaries with total mass (in the source rest frame) as small as  $10^4 M_{\odot}$  and up to  $10^7 M_{\odot}$ , out to a redshift as remote as z ~ 20 .
- NGO will measure red-shifted mass to an unprecedented accuracy, up to the 0.1 % 1 % level, whereas absolute errors in the spin determination are expected to be in the range 0.01 to 0.1 dimensionless units.
- Luminosity distance to 1-50 %. Sky location for brightest  $\approx$ 3° -10°
- Final merger in GR leads to a single ringing Kerr black hole characterized by its mass and spin. Detecting two quasi-normal modes checks whether the final object indeed is described only by two parameters in accord with the "no-hair" theorem of GR.





## Extreme Mass Ratio inspiral



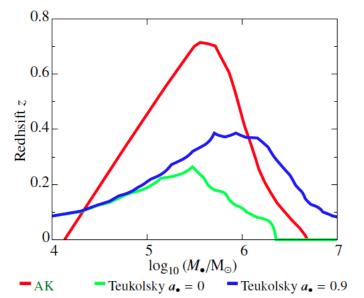


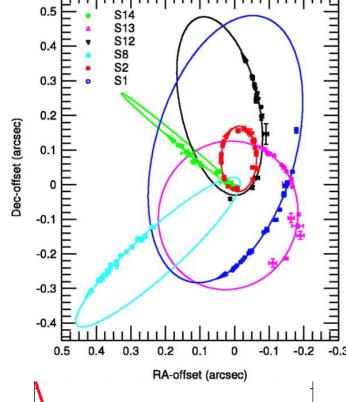
- Stellar mass black-hole into a Massive black-hole
- SNR 20 up to  $z \approx 0.7$

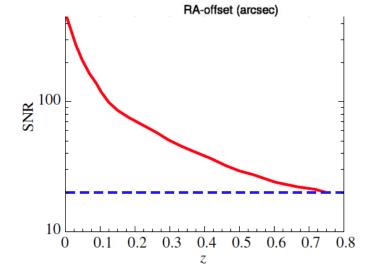
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• Rate in 2 years (2< <50)

Waveform model	wi	Analytic Kludge		
	$a_{\bullet} = 0$	$a_{\bullet} = 0.5$	$a_{\bullet} = 0.9$	
Number of events	30	35	55	50





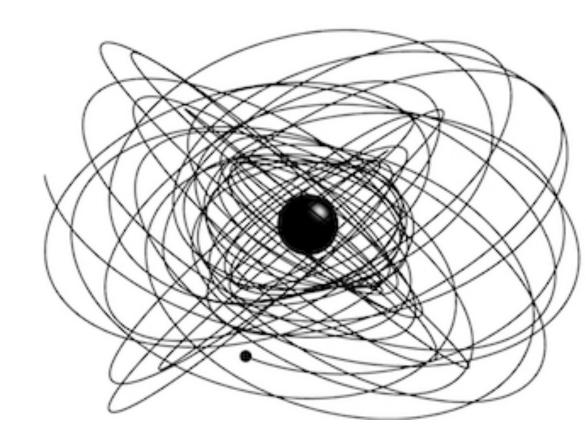




## Extreme Mass Ratio inspiral



- NGO observes ~  $10^5$  final cycles of the inspiral of a stellar mass object into a black hole with mass  $10^5 M_{\odot} 10^6 M_{\odot}$  up to z ~ 0.7.
- Give precisely the parameters of the central object including mass, spin (0.1-0.01 % accuracy) and quadrupole moment (0.001  $M_{\odot}^{3}G^{2}/c^{4}$ ).
- Allows to check whether it is consistent with a GR Kerr black hole.





## Science objectives



#### • Survey compact stellar-mass binaries and study the structure of the Galaxy

- Elucidate the formation and evolution of Galactic stellar-mass compact binaries and thus constrain the outcome of the common envelope phase and the progenitors of (type Ia) supernovae.
- Determine the spatial distribution of stellar mass binaries in the Milky Way
- Improve our understanding of white dwarfs, their masses, and their interactions in binaries, and enable combined gravitational and electromagnetic observations

#### • Trace the formation, growth and merger history of massive black holes.

- − Trace the formation, growth and merger history of massive black holes with masses  $10^5 \, M_\odot 10^7 \, M_\odot$  during the epoch of growth of quasi-stellar objects and widespread star formation (0 < z < 5) through their coalescence in galactic halos.
- − Capture the signal of coalescing massive black hole binaries with masses  $2 \times 10^4 \, M_{\odot} 10^5 \, M_{\odot}$  in the range of 5 < z < 10 when the universe is less than 1 Gyr old.

#### • Explore stellar populations and dynamics in galactic nuclei

- Characterize the immediate environment of massive black holes in z < 0.7 galactic nuclei from extreme mass ratio capture signals.
- Discovery of intermediate-mass black holes from their captures by massive black holes.

#### • Confront General Relativity with observations

- Detect gravitational waves directly and measure their properties precisely
- Test whether the central massive objects in galactic nuclei are consistent with the Kerr black holes of General Relativity.
- Perform precision tests of dynamical strong-field gravity.

#### Probe new physics and cosmology with gravitational waves

- Measure the spectrum of cosmological backgrounds, or set upper limits on them in the  $10^{-4}$  Hz  $10^{-1}$  Hz band.
- Search for gravitational wave bursts from cosmic string cusps and kinks.
- Cosmography with standard sirens Maritime Institute 20/12/2011

#### NGO Science Working Team

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